

ULDB Demonstration Program: Balloon Craft

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The Challenge

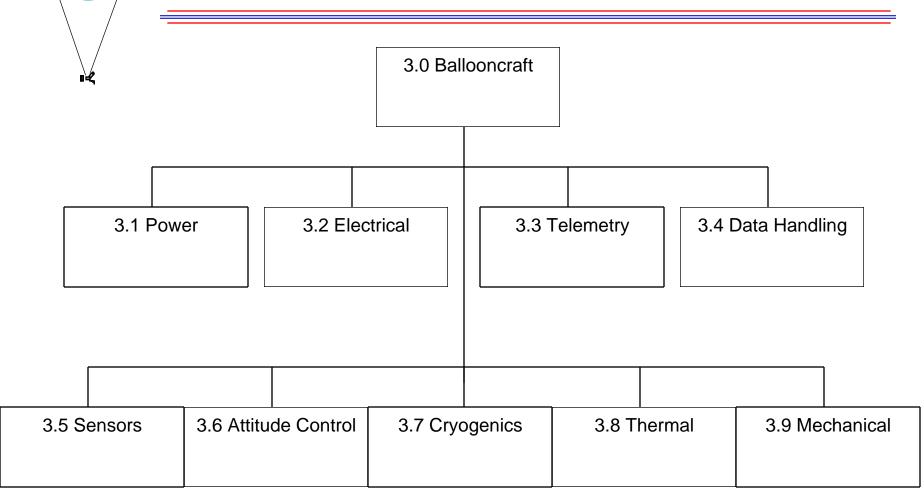
- Provide comprehensive set of Flight and Ground Support Systems
 - Mission Operations Requirements
 - ULDB Science Requirements
 - TIGER Instrument Requirements

The Approach

- GSFC Civil Servant Development Team
 - Requirements Definition
 - Trade Studies/Design Concepts
 - Design Reviews
 - Engineering Test Flight
 - Demo 2000 Flight with TIGER science instrument



ULDB Ballooncraft Development Team





Power Subsystem

- Requirement
 - Scalable, low weight power subsystem capable of supporting up to 1000 watts continuous power consumption
 - 12,000 watt-hours daytime, 12,000 watt-hours nighttime
 - Demo 2000 requirement
 - 4800 watt-hours daytime, 7200 watt-hours nighttime
- Design
 - Deployed amorphous silicon flexible thin film PV panels
 - Solar tracking in azimuth
 - Nickel Metal Hydride (NiMH) battery power storage



• "Telemetry Subsystem

- Requirement
 - Global downlink of science data at continuous rate > 9kbps
 - Redundant low rate global downlink of science and operations housekeeping data
 - Redundant low rate global uplink of science and operations commands
- Design
 - TDRSS with High Gain Antenna (HGA)
 - New contract with Motorola for "balloon-class" transponder
 - » Lower cost version of 4th generation space qualified unit
 - Wallops in-house HGA
 - Bit rate capability >100 kbps MA, >300 kbps SSA
 - Low rate global communications
 - Inmarsat and ARGOS flight proven
 - Iridium and Orbcomm under development



Flight Data Processing and Data Storage

- Requirement
 - Acquire, process and execute commands
 - Acquire, process, format data
 - On-board storage 10 GB required, 50 GB desired
- Design
 - COTS PC-104 based 486-class flight processors
 - COTS real-time operating system
 - Multiple data interfaces
 - RS-232, Analog, Discrete Digital, MIL-STD-1553
 - Data storage using sealed hard disk drives
 - 18 GB per drive current density
 - 36 GB per drive in 1999



Attitude and Position Determination

- Requirement
 - Determine location and altitude during flight
 - Determine absolute azimuth for TDRSS HGA and science data analysis
- Design
 - COTS Attitude Determination GPS unit
 - Provides time reference
 - Provides 3-D location with high accuracy
 - Provides azimuth determination
 - 4 antennas 1 meter apart
 - One degree accuracy



Attitude Control

- Requirement
 - Sun-tracking of solar panels
- Design
 - WFF-developed azimuth positioner
 - Points entire gondola
 - < 5 watts
 - Better than 5 degree accuracy
 - Details in later presentation

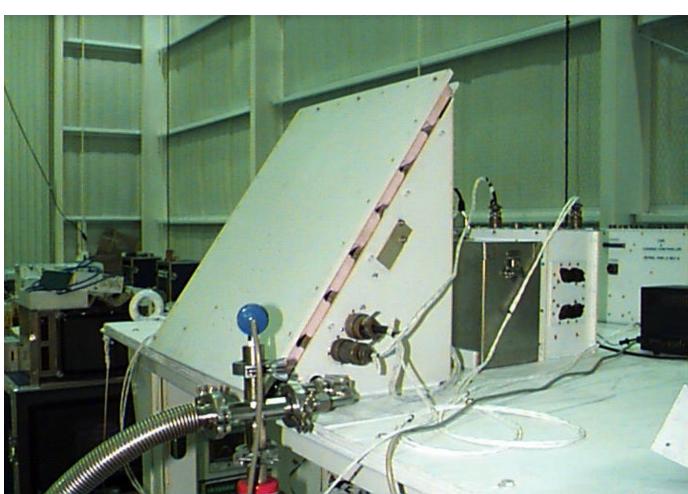


Cryocooling

- Requirement
 - Balloon flight-qualify Sunpower Model M77 stirling cycle cryocooler for future ULDB instrument use for detector cooling
- Design
 - Sunpower Cryocooler test flight June 1998 on LDB flight from Alaska
 - WFF in-house housing, thermal control system, vibration monitoring system
 - GSFC/Greenbelt 552 has tested this unit and developed vibration compensation circuitry
 - Test flight demonstrated successful operation at balloon float altitudes



ULDB Cryocooler Test Flight

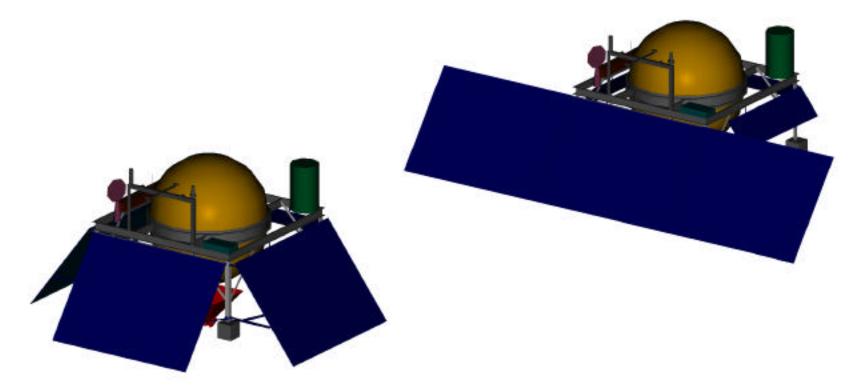




Thermal and Mechanical

- Requirement
 - Perform Thermal Analysis and develop thermal control systems to maintain ballooncraft within operating limits
 - Design and provide ballooncraft structure, enclosures and mechanisms
- Design
 - In-house thermal analysis and control
 - Application of passive control techniques first priority
 - In-house mechanical design of gondola structure, PV panel structure and deployment mechanism, Command and Data Module, etc.







Operations Control Center

- Requirement
 - Support flight operations via line of sight and global communications
 - Support off-site instrument data delivery and command receipt
- Design
 - Combination of GOTS and COTS on workstations
 - GOTS Integrated Test & Operations System (ITOS)
 - » Used on SMEX, SMEX Lite, SPARTAN
 - » ITOS to be enhanced to support remote data monitoring and control
 - COTS products for enhanced operations displays